



Ring Laser Gyroscope Two-Axis Indexing Inertial Navigation System

Ring Laser Gyroscope Two-Axis Indexing Inertial Navigation System-D90C is an inertial navigation system based on ring laser gyroscope and quartz flexible accelerometer. Can be combined with altimeter, GNSS, airspeed meter and other auxiliary equipment, can provide air carrier, ground carrier and other navigation information, such as speed, position and attitude.

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Technical features

- Two-axis indexing mechanism for error mitigation
- High accuracy ring laser gyro and quartz accelerometer
- Optional static or moving base self-alignment
- Error parameters calibration and compensation in full temperature range
- Optional diverse input interfaces for GNSS/Odometer/DVL
- Configurable navigation modes
- Excellent environmental suitability
- Military standards

Typical application scenarios

Under-sea vehicle navigation
Positioning and north-finding for land vehicle
Stabilization and control for moving carrier
Attitude measurement for demanding applications

Main functions of the system

It has the function of outputting information such as carrier position, heading, attitude angle, angular rate and speed in real time;

It has working modes such as pure inertial navigation and INS/GNSS (including Beidou) integrated navigation;

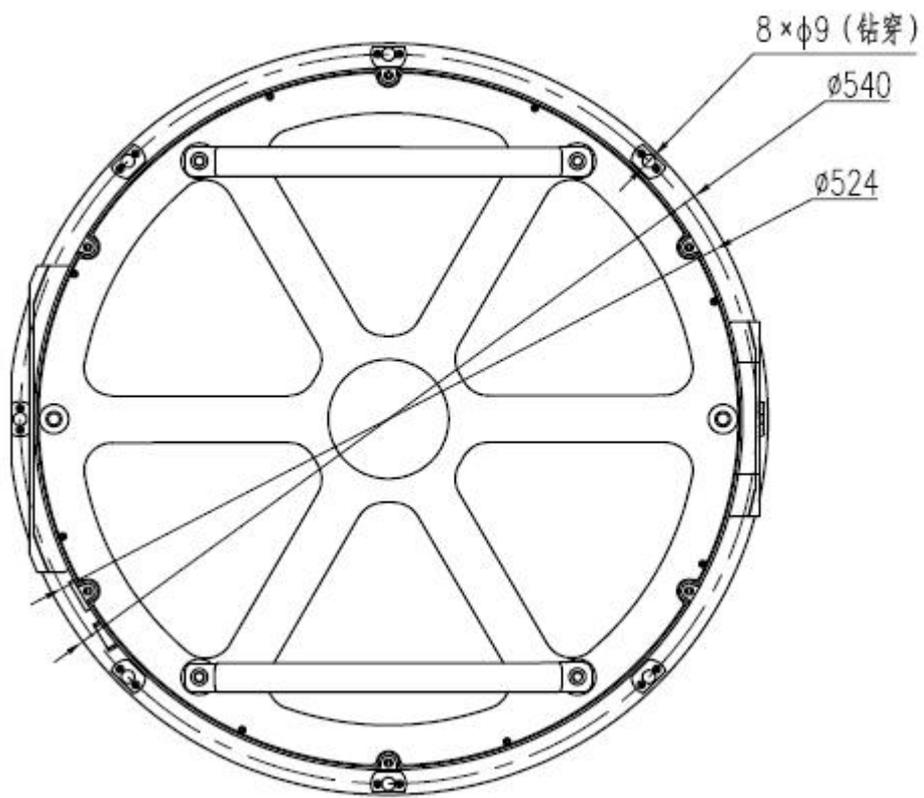
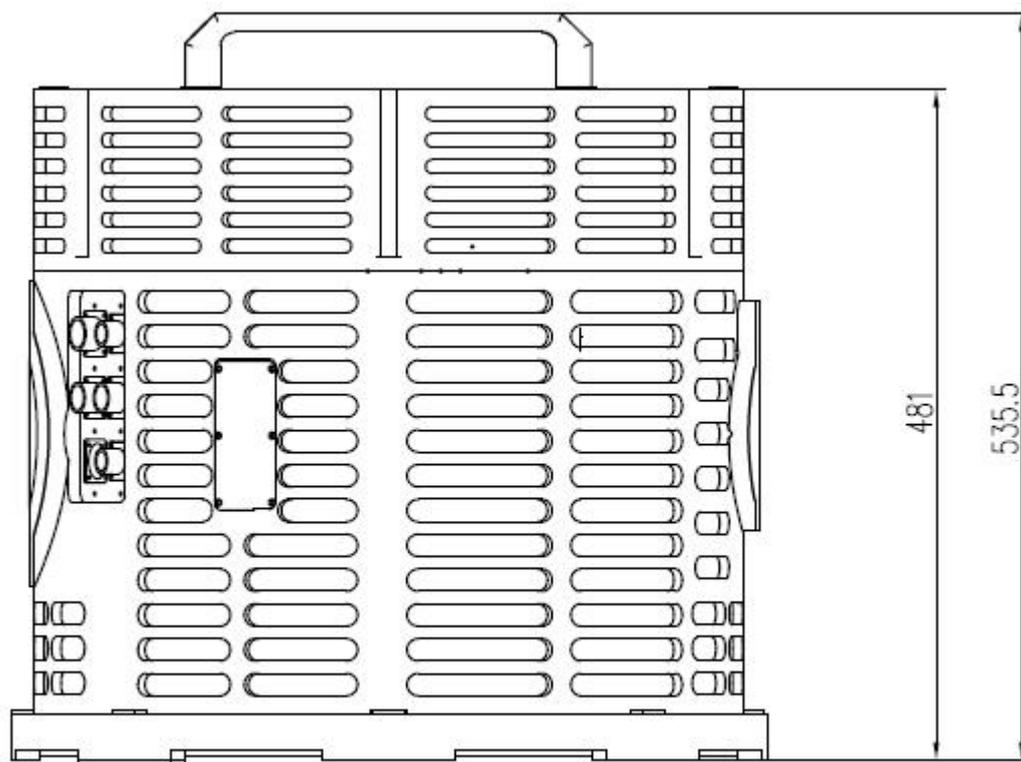
Possess the function of receiving satellite navigation information provided by external time system frequency standard equipment;

It has the function of ground self-alignment and supports the function of air alignment;

It has functions such as power-on self-test, periodic self-test, status report, installation error compensation, and non-volatile storage.

System composition and dimensions

The whole system is composed of two parts: the inertial navigation main instrument and the inertial navigation main instrument bracket. Among them, the external dimensions of the main instrument are as follows:



Main technical indicators of the system

System accuracy System Accuracy	Pure Inertial Navigation/Pure Inertial Navigation		2.0nmile/5d, PEAK
	Integrated Navigation/Navigation with GNSS		≤5m , 1σ _
	Heading angle /Heading		0.01°, RMS
	Horizontal attitude (roll and pitch) Horizontal Attitude (roll & pitch)		0.005°, RMS
	Pure Inertial Velocity		1m/s , RMS
	GNSS Integrated navigation Velocity		0. 1 m/s , RMS
Indicators of inertial devices Gyro and Accelerometer Parameters	laser gyroscope _ Gyroscope	Range/Range	± 6 00 deg/s
		Bias Stability	≤0.02 deg/h, 1σ
		Zero bias repeatability Bias Repeatability	≤0.02 deg/h, 1σ
		Scale Factor non-linearity	1 ppm
	Accelerometer Accelerometer	Range/Range	± 15g _
		Bias Stability	≤10μg , 1σ
		Zero bias repeatability Bias Repeatability	≤10μg , 1σ
		Scale Factor non-linearity	15 ppm
align time Alignment Time	Cold Start		≤ 10 min
	Re-Start		≤15min
	Air/In-Flight Start		more than 10h
Interface Features interface	Supply voltage/Voltage		18~36VDC
	Power Consumption		≤ 40W @ 24VDC
	Electrical interface/Electrical		RS232 × 2 RS422 × 3 CAN × 2 Ethernet × 1 1pps × 1
	Data Update Rate (configurable)		200Hz@115.2kbps _ _ _

Use environment Environmental	Operating Temperature	-40°C~+65°C
	Storage temperature/Storage Temperature	-55°C~+85°C
	Use Altitude/Altitude	20000m
	Humidity	95% (+25°C)
	Vibration/Vibration	5g @ 20~2000Hz
	Shock/Shock	40 g, 11 ms, 1/2 Sine
Physical properties Physical	Dimensions/ Size (L*W*H)	540 x 536mm _
	Weight/ Weight	55 kg

Note: The structure can be customized according to the user's requirements.